

AEROSPACE MATERIAL SPECIFICATIONS

AMS 6485A

SOCIETY OF AUTOMOTIVE ENGINEERS, Inc. 485 Lexington Ave., New York 17, N.Y.

Issued 11-1-59
Revised 6-30-62

STEEL
5Cr - 1.3Mo - 0.5V (0.38 - 0.43C)

1. ACKNOWLEDGMENT: A vendor shall mention this specification number and its revision letter in all quotations and when acknowledging purchase orders.

2. FORM: Bars, forgings, and forging stock.

3. APPLICATION: Primarily for parts requiring relatively high levels of strength, fatigue resistance, toughness, ductility, and thermal stability for operation between -100 F (-75 C) and 1000 F (540 C), and where such parts may require welding.

4. COMPOSITION:

Carbon	0.38 - 0.43
Manganese	0.20 - 0.40
Silicon	0.80 - 1.00
Phosphorus	0.020 max
Sulfur	0.020 max
Chromium	4.75 - 5.25
Molybdenum	1.20 - 1.40
Vanadium	0.40 - 0.60

4.1 Check Analysis: Composition variations shall meet the requirements of the latest issue of AMS 2248; check analysis tolerance for vanadium shall be 0.04 under min or over maximum.

5. CONDITION:

5.1 Bars: In a machinable condition and hot finished having hardness not higher than ϕ Brinell 235 or equivalent, except that bars ordered cold finished may have hardness as high as Brinell 255 or equivalent.

5.2 Forgings: As ordered.

5.3 Forging Stock: As ordered by the forging manufacturer.

6. TECHNICAL REQUIREMENTS:

6.1 Decarburization:

6.1.1 Bars ordered ground, turned, or polished shall be free from decarburization.

6.1.2 Allowable decarburization of bars ordered for redrawing or forging, or to specified microstructural requirements, shall be as agreed upon by purchaser and vendor.

Section 8.3 of the SAE Technical Board rules provides that: "All technical reports, including standards approved and practices recommended, are advisory only. Their use by anyone engaged in industry or trade is entirely voluntary. There is no obligation to adhere to an SAE standard or recommended practice, and no commitment on the part of the SAE Board or its committees to investigate or consider patents which may apply to the subject matter. Prospective users of the reports are responsible for protecting themselves against infringement of patents."

- 6.1.3 Decarburization of bars to which 6.1.1 or 6.1.2 is not applicable shall be not greater than the following:

Nominal Diameter or Distance Between Parallel Sides Inches	Depth of Decarburization Inch
0.375 and under	0.010
Over 0.375 to 0.500, incl	0.015
Over 0.500 to 0.625, incl	0.020
Over 0.625 to 1.000, incl	0.025
Over 1.000 to 2.000, incl	0.035
Over 2.000 to 3.000, incl	0.048
Over 3.000 to 4.000, incl	0.062
Over 4.000 to 5.000, incl	0.094
Over 5.000	0.125

- 6.1.4 Unless otherwise agreed upon by purchaser and vendor, decarburization shall be measured by the microscopic method, or by Rockwell Superficial 30-N scale hardness method, or equivalent hardness testing method, on hardened specimens. Depth of decarburization, when measured by a hardness method, is defined as the depth below which there is no further increase in hardness; such measurements shall be far enough away from any adjacent surface to be uninfluenced by any decarburization or lack of decarburization thereon.

- 6.1.4.1 When determining the depth of decarburization, it is permissible to disregard local areas provided the decarburization of such areas does not exceed the limits above by more than 0.005 in. and the width is 0.065 in. or less.

- 6.2 Properties After Heat Treatment: Specimens austenitized by heating to $1850\text{ F} + 25$ ($1010\text{ C} + 14$), holding at heat for 15 - 45 min., and cooling in air to room temperature and then tempered three times by heating to not lower than 1000 F (538 C), holding at heat for 2 - 3 hr, and cooling in air shall conform to the following requirements.

- 6.2.1 Longitudinal Tensile Properties: The following requirements apply to specimens taken from bars and forging stock 25 sq in. and under in cross sectional area and from forgings with axis approximately parallel to the forging flow lines; in addition, specimens from coupons of stock over 25 sq in. in cross sectional area forged to 25 sq in. in area shall be capable of meeting these requirements.

Tensile Strength, psi	260,000 min
Yield Strength at 0.2% Offset or at 0.0181 in. in 2 in. Extension Under Load ($E = 30,400,000$), psi	215,000 min
Elongation, % in 4D	8 min
Reduction of Area, %	30 min

6.2.2 Transverse Tensile Properties: The following requirements apply to specimens taken from bars and forging stock over 25 to 256 sq in., incl, in area, selected and prepared in accordance with the latest issue of AMS 2310 and heat treated as in 6.2:

Tensile Strength, psi	260,000 min
Yield Strength at 0.2% Offset at 0.0181 in. in 2 in. Extension Under Load ($E = 30,400,000$), psi	215,000 min
Reduction of Area, %	
Cross Sectional Area, sq in.	Average Minimum
Over 25 to 75, excl	15 6
75 to 100, incl	10 6
Over 100 to 150, incl	-- 5
Over 150 to 225, incl	-- 4
Over 225 to 256, incl	-- 3

6.2.3 Hardness: Rockwell C 50 - 56 or equivalent, but hardness shall not be cause for rejection if tensile properties are met.

6.2.4 Grain Size: Predominantly 7 or finer with occasional grains as large as 5 permissible, ASTM E112-61T; the procedure used shall be as agreed upon by purchaser and vendor.

6.2.5 Longitudinal Tensile Properties at 1000 F (537.8 C): Test specimens heated to 1000 F + 10 (537.8 C + 5.6), held at 1000 F + 10 (537.8 C + 5.6) for 30 min., and tested at 1000 F \pm 5 (537.8 C \pm 2.8) shall be capable of meeting the following requirements. These properties apply when the rate of strain is maintained at 0.003 - 0.007 in. per in. per min. through the yield strength and then is increased so as to produce failure in approximately one additional minute.

Tensile Strength, psi	175,000 min
Yield Strength at 0.2% Offset or at 0.0157 in. in 2 in. Extension Under Load ($E = 23,000,000$), psi	135,000 min
Elongation, % in 4D	10 min
Reduction of Area, %	35 min

7. QUALITY: Steel shall be aircraft quality and shall conform to the requirements of the latest issue of AMS 2301. Material shall be uniform in quality and condition, sound, and free from foreign materials and from internal and external imperfections detrimental to fabrication or to performance of parts.

8. TOLERANCES: Unless otherwise specified, tolerances shall conform to all applicable requirements of the latest issue of AMS 2251 for hot finished; for bars ordered cold finished and for all hexagons, tolerances for cold finished shall apply.

9. REPORTS:

9.1 Unless otherwise specified, the vendor of the product shall furnish with each shipment three copies of a report of the results of tests for chemical composition of each heat in the shipment and the results of tests on each size from each heat to determine conformance to the technical requirements of this specification. This report shall include the purchase order number, heat number, material specification number, size, and quantity from each heat. If forgings are supplied, the part number and size of stock used to make the forgings shall also be included.